

工業技術研究院

Industrial Technology
Research Institute

3D Pitot Tube Measurements and Calibration in the Wind Tunnel

Center for Measurement Standards
Industrial Technology Research Institute
Taiwan, R.O.C.

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April 20, 2015



About ITRI

- Founded in 1973



● Total Staff : 5,782

● Ph.D. : 1,295

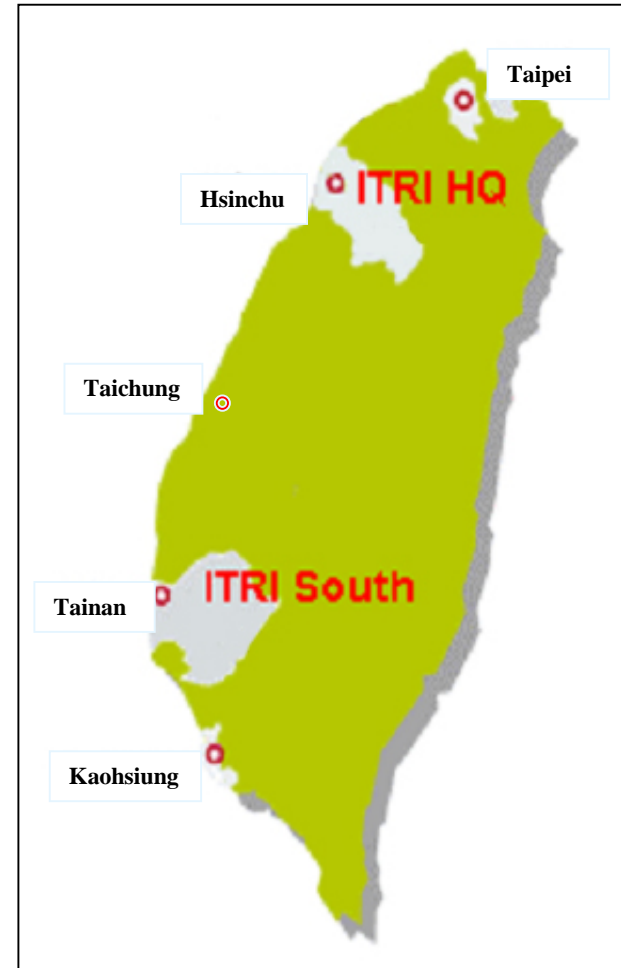
● Total Patents : 16,732

● Start-Ups : 171

About ITRI

- A not-for-profit non-government R&D organization

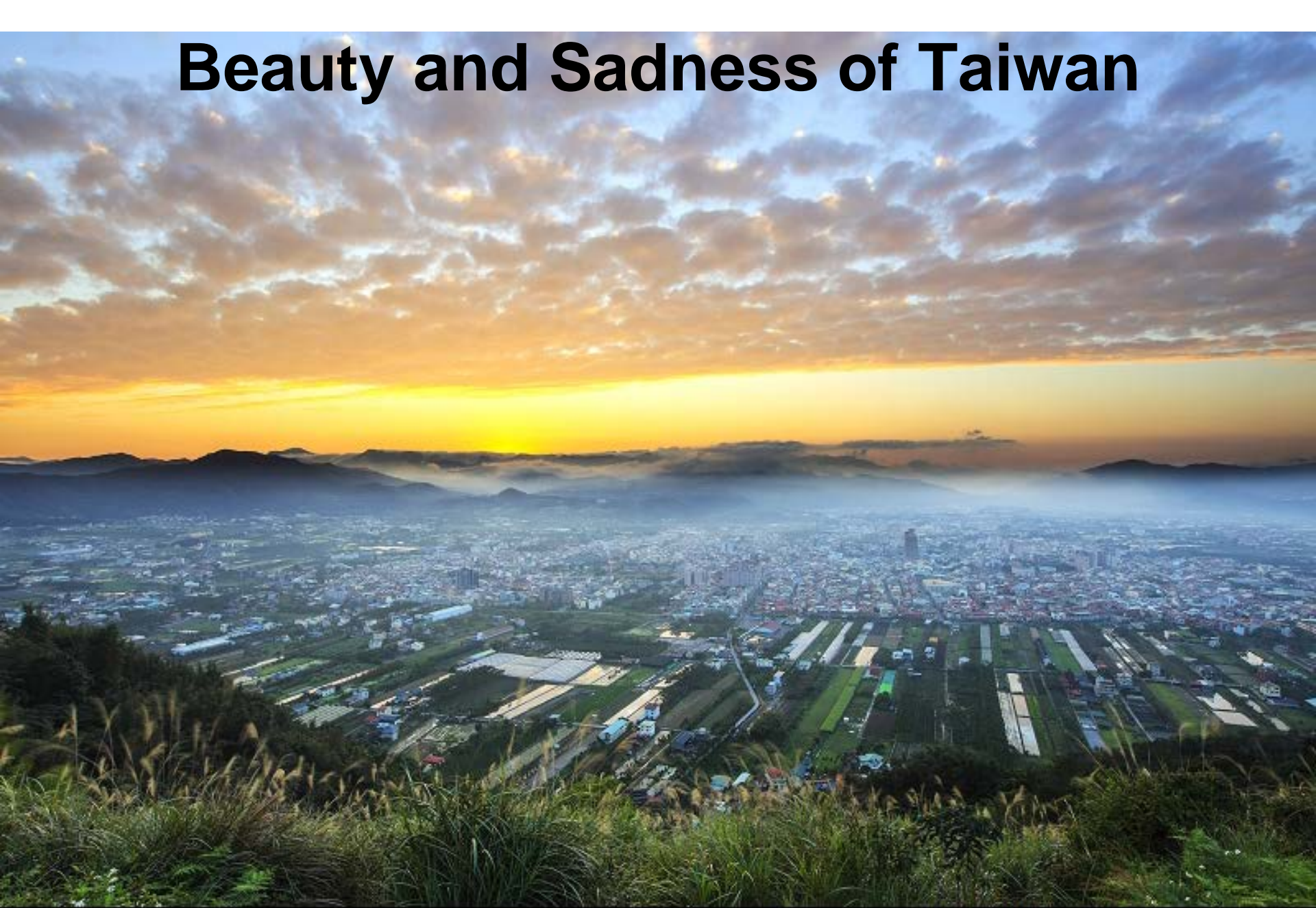
- ④ To create economic value through innovation and technology R&D
- ④ To spearhead the development of emerging new industry
- ④ To enhance the competitiveness of Taiwanese industries in the global market



Contents

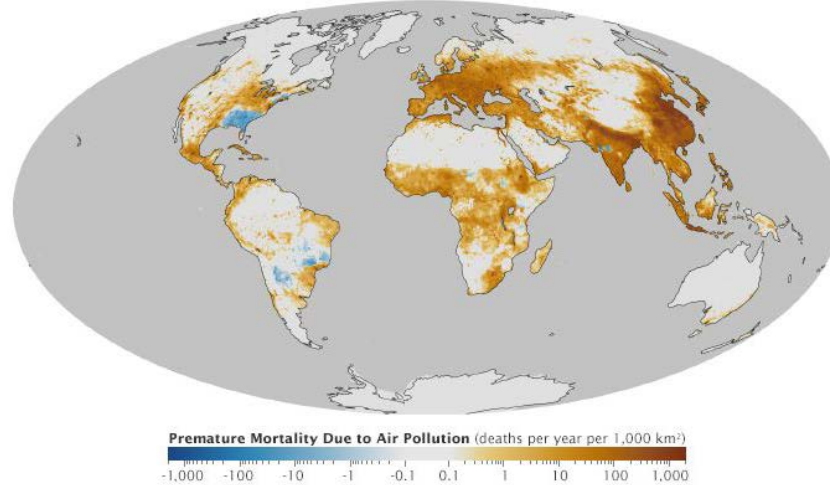
- Introduction
- Technology Needed for Smokestack Flow Measurements
- Characterization of Pitot Tubes
- Calibration Facilities
- Calibration Data Analysis
- Future Work

Beauty and Sadness of Taiwan



小羊比遊記 · <http://yangbi.idv.tw>

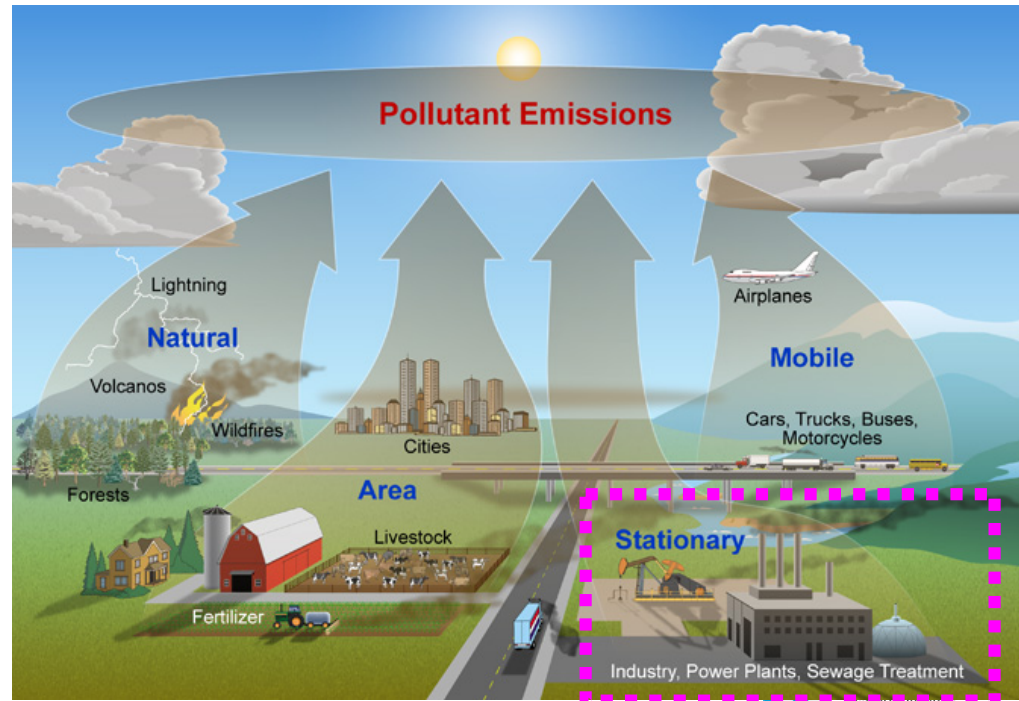
Introduction



Where on earth are you most likely to die early from air pollution?

Stationary Source Emissions

in air quality terminology is any fixed emitter of air pollutants, such as fossil fuel burning power plants, petroleum refineries, petrochemical plants, food processing plants and other heavy industrial sources



Technology Needed for Smokestack Flow Measurements



Touch panel manufacturers



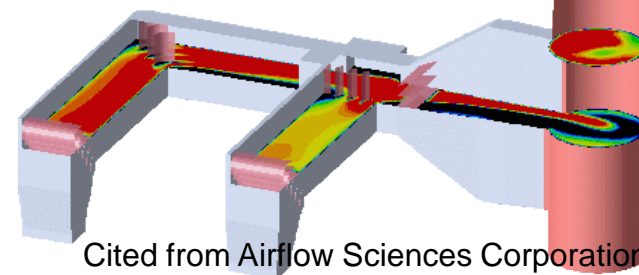
Waste incineration plants



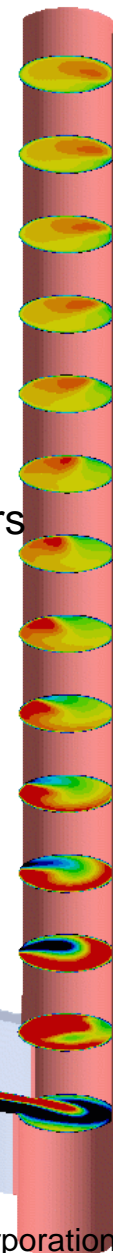
Semiconductor manufacturers

Technical challenges

- Swirl and inhomogeneous flow
- Complicated compositions
- Location of measurements
- Calibration of instruments

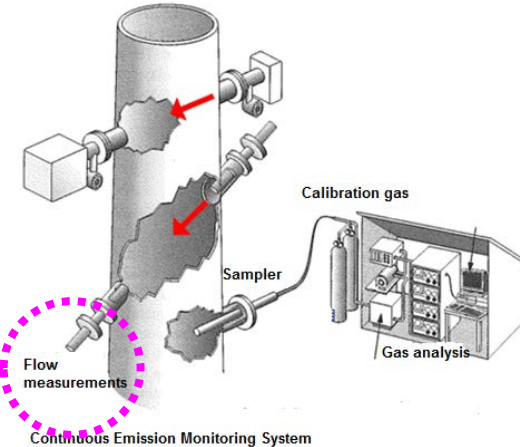
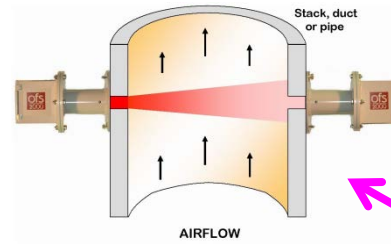


Cited from Airflow Sciences Corporation



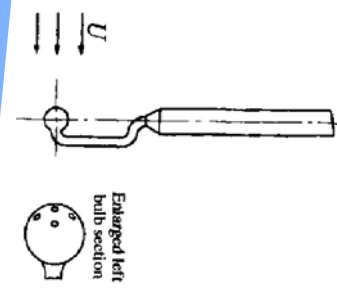
Characterization of Pitot Tubes

- Introduction



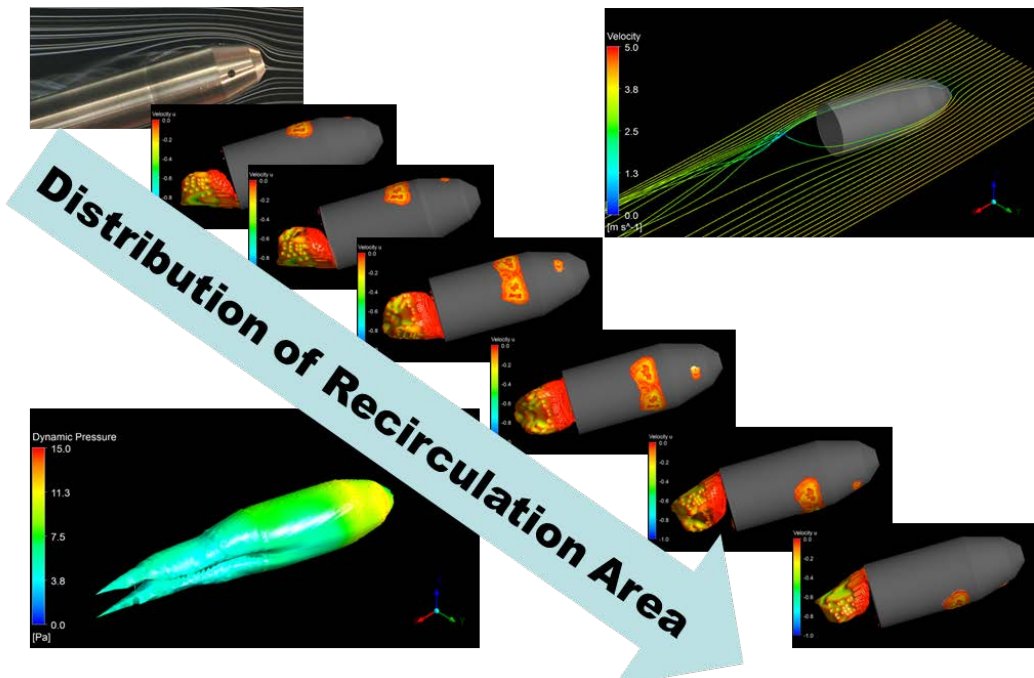
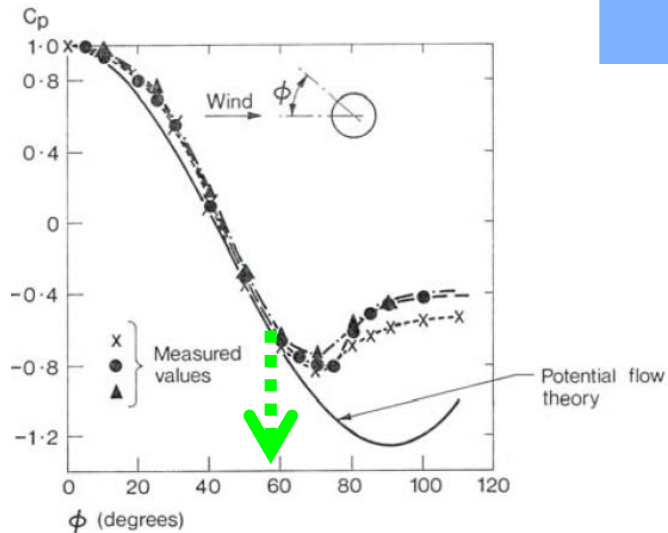
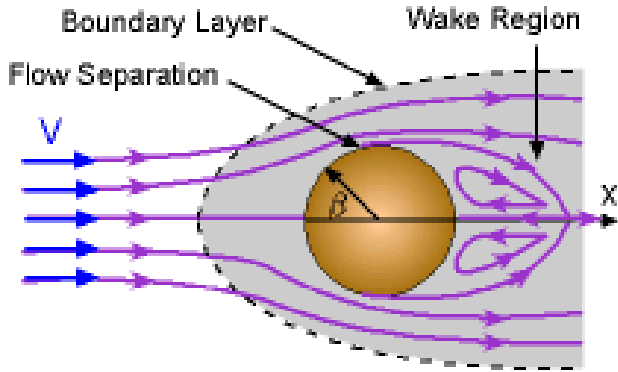
EPA Methods 2F, 2G, and 2H (40 CFR Part 60, Appendices A-1 and A-2)

- ◆ Method 2 is fine for situations where there is a straight forward flow profile, however it is prone to bias if cyclonic flow and wall effects are ignored
- ◆ Method 2G is used for accurate velocity and volumetric flow rate measurements when stack gas has significant yaw angle; it may be conducted using either a Type S or 3-D probe
- ◆ Method 2F is used for accurate measurements when stack gas has significant yaw and pitch angle; it must be conducted using a three-dimensional (3-D) DAT or spherical probe
- ◆ Method 2H (or CTM-041 for rectangular ducts) is used for accurate measurements by accounting for velocity drop-off near the stack or duct wall



Characterization of Pitot Tubes

- Flow visualization of pitot probes by CFD



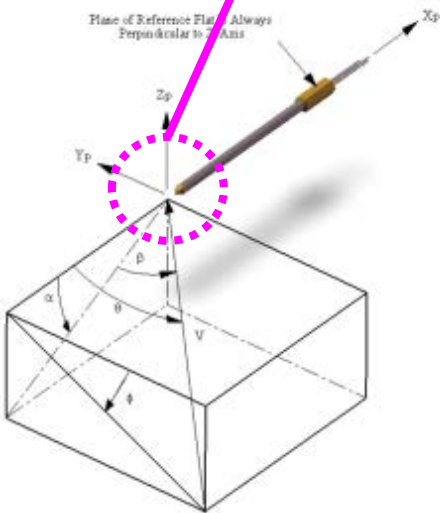
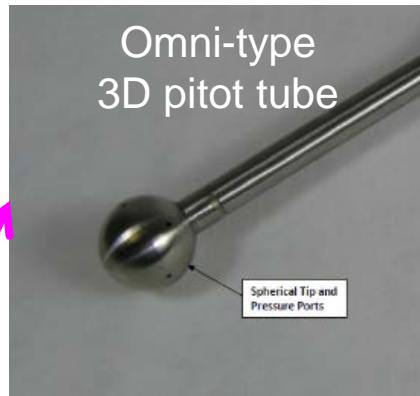
Governing equation: $\frac{p}{\rho} + \frac{V^2}{2} + gz = \text{constant}$

- Assumptions:**
- (1) Steady flow.
 - (2) Incompressible flow.
 - (3) Flow along a streamline.
 - (4) Frictionless deceleration along stagnation streamline.

$$V = \sqrt{\frac{2(p_0 - p)}{\rho_{\text{air}}}}$$

Characterization of Pitot Tubes

- Design of 3D pitot tubes



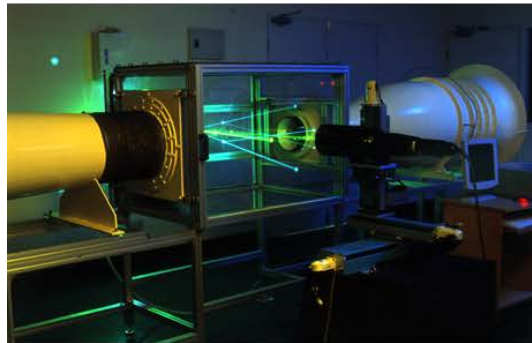
| Geometry and Construction | | Measurement Accuracy (w/Aeroprobe Calibration) | |
|---------------------------|--|---|---|
| Probe Geometry | Straight, L-Shaped | Flow Angles | $< 1^\circ$ |
| Number of Holes | 12 | Total Flow Velocity | $< 1\%*$ |
| Tip Geometry | Spherical | Required Auxiliary Data** | Reference Pressure, Total Temperature |
| Tip Diameter | 6.35 mm, 9.53 mm | Flow Angle of Receptivity | 120° |
| Material | All-Stainless Steel Construction | Calibration Flow Speeds | 5 m/s to 315 m/s (Mach = 1.0) |
| Pneumatic Connection | Tygon R3603 Formulation, 1/32" ID, 3/32" OD Standard for Exit Tubing of 0.89 mm – 1.6 mm (0.035" – 0.063") OD. | Pressure Data Reduction | Omnipro Software |
| | | Frequency Response | Low, Best for Determining Time-Averaged Flows |
| Mounting | Hex Prism (standard) | Media | Non-Reactive Gases |
| Probe Reference | Flat on Hex Mount with "R" | *Utilizing 0.1% Accurate Pressure Sensors Properly Rated for Flow Speed | |
| Flow Temp. Limits | $0^\circ\text{C} - 150^\circ\text{C}$ | **For Most Accurate Compressible P-V Reduction | |

Calibration Facilities

- Calibration system construction at CMS

3D Motion → 3D Measurements

3D Calibration is needed !!



Wind tunnel



3D pitot tube



Traversing control system

- Pitch/Yaw angle control
- Turbulence intensity control



Pressure scanner

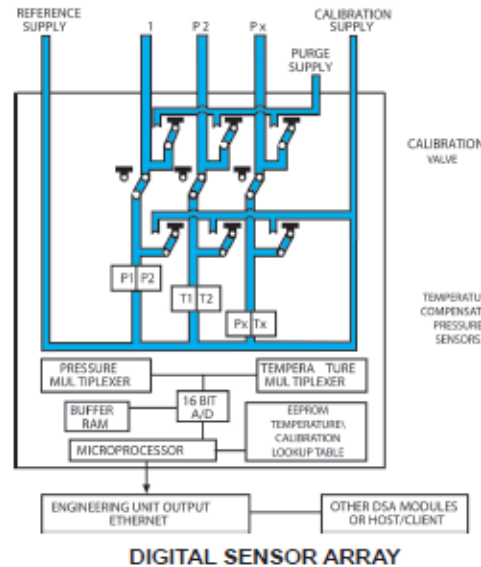
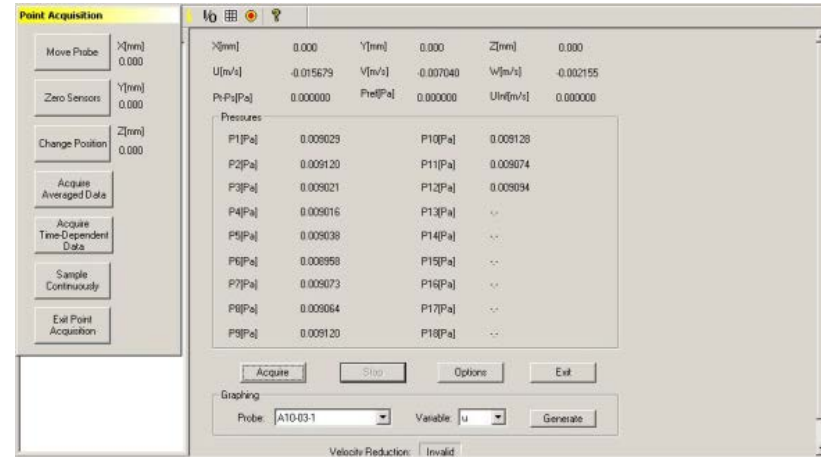
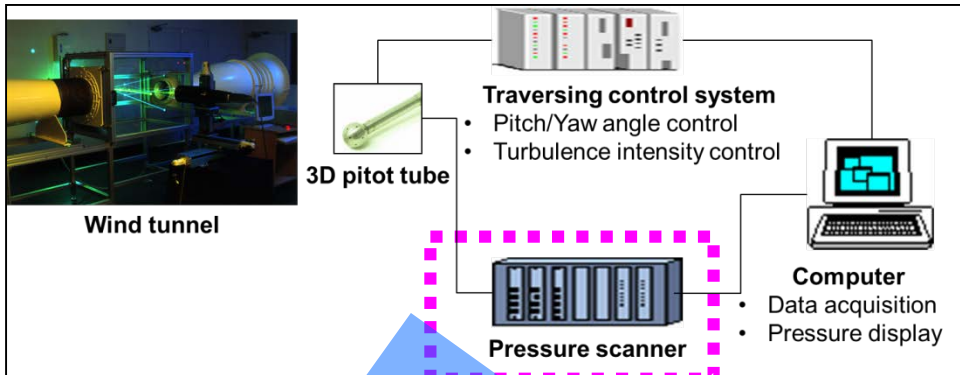


Computer

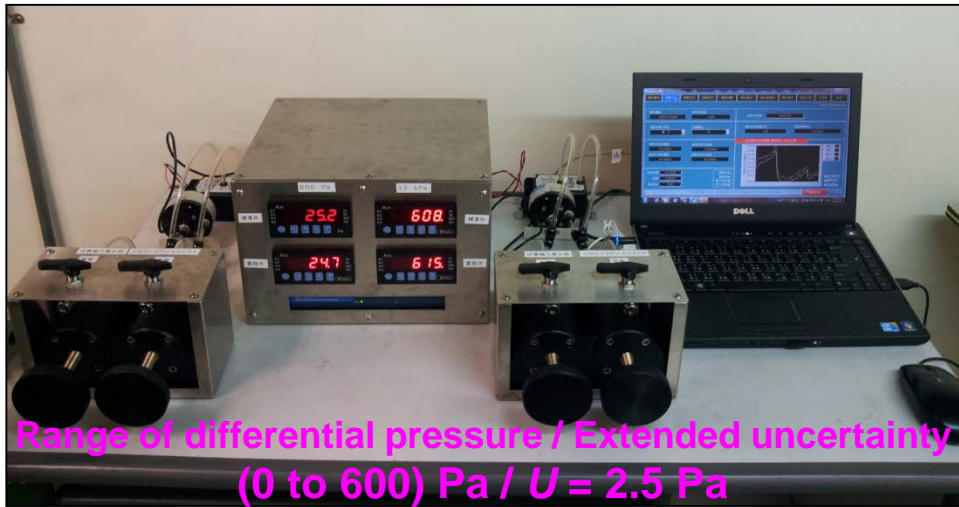
- Data acquisition
- Pressure display

Calibration Facilities

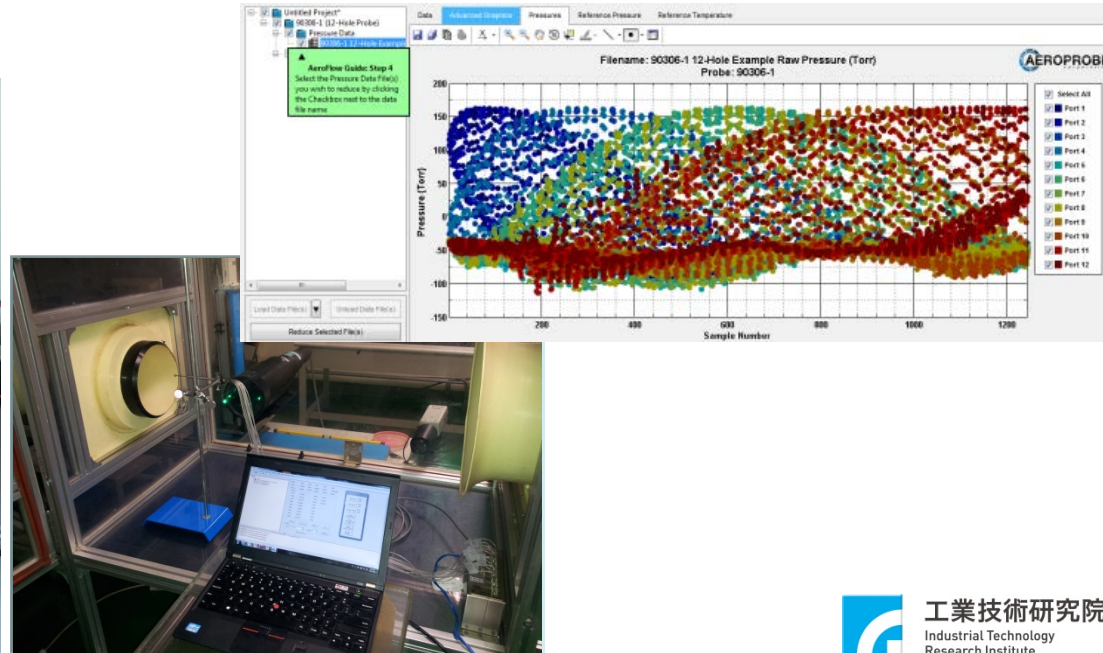
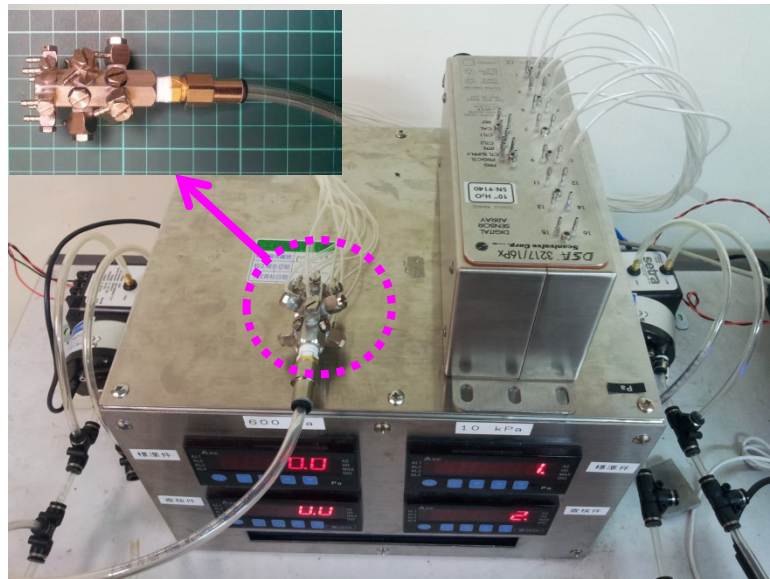
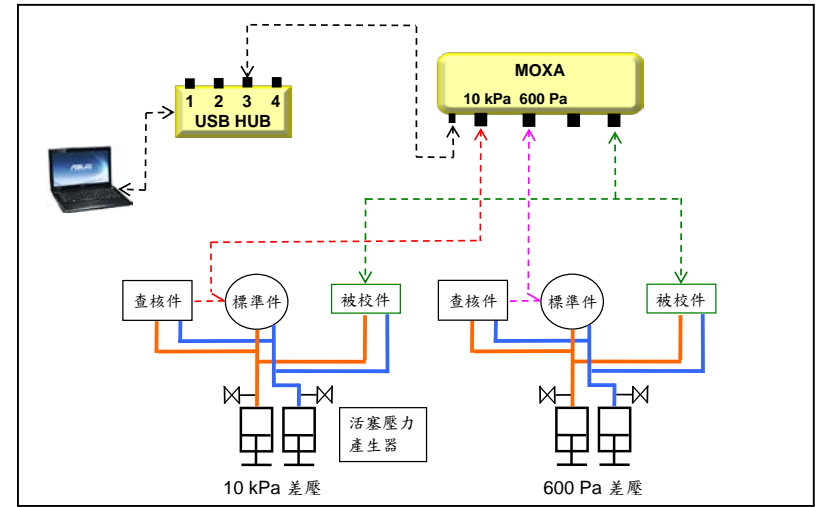
- Pressure measurements and calibration



| Sensor Pressure Range | Static Accuracy (% F.S.) |
|--|--------------------------|
| ±10 inch H ₂ O (2.5 kPa) ² | ±.20% |
| ±1, ±2.5 psid | ±.12% |
| ±5 to 500 psid | ±.05% |
| ±501 to 750 psid | ±.08% |

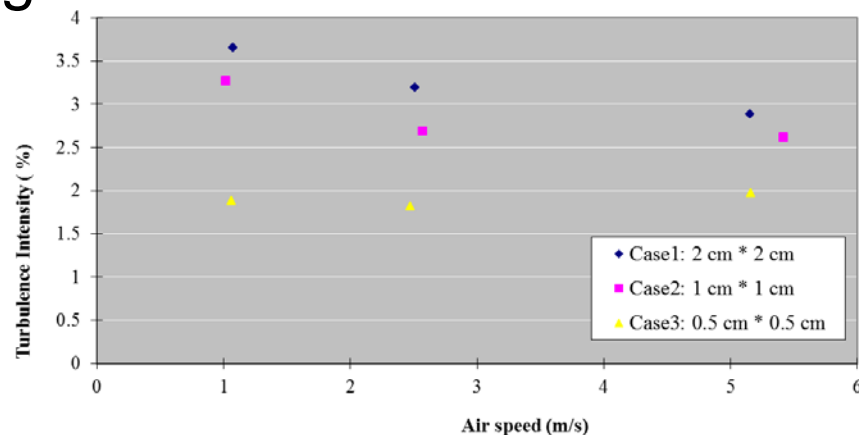
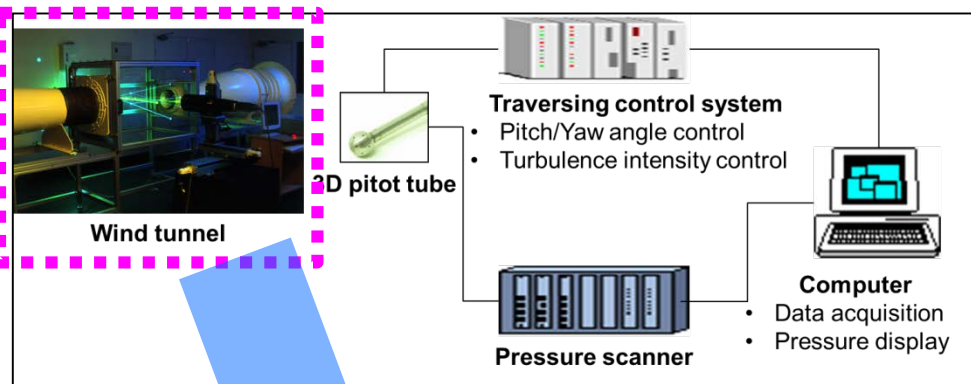


Piston-type Pressure Calibrator

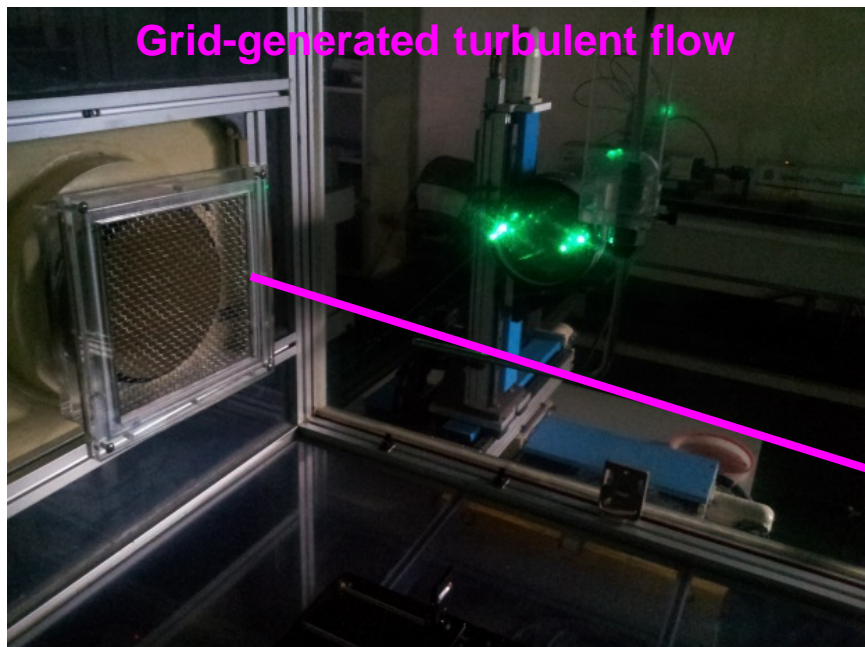


Calibration Facilities

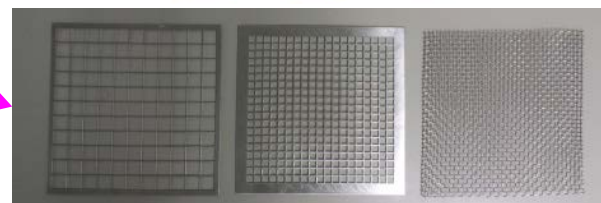
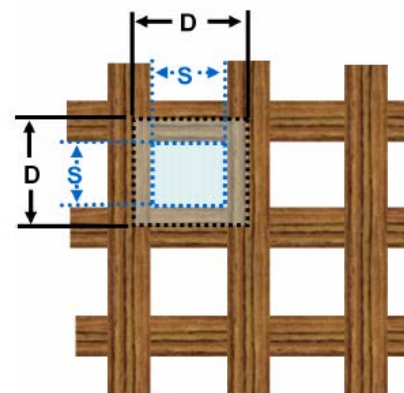
- Turbulence generation



- Without grids: T.I.= (0.7 to 1.4) %
- With grid No.4: T.I.= (1.7 to 4.4) %

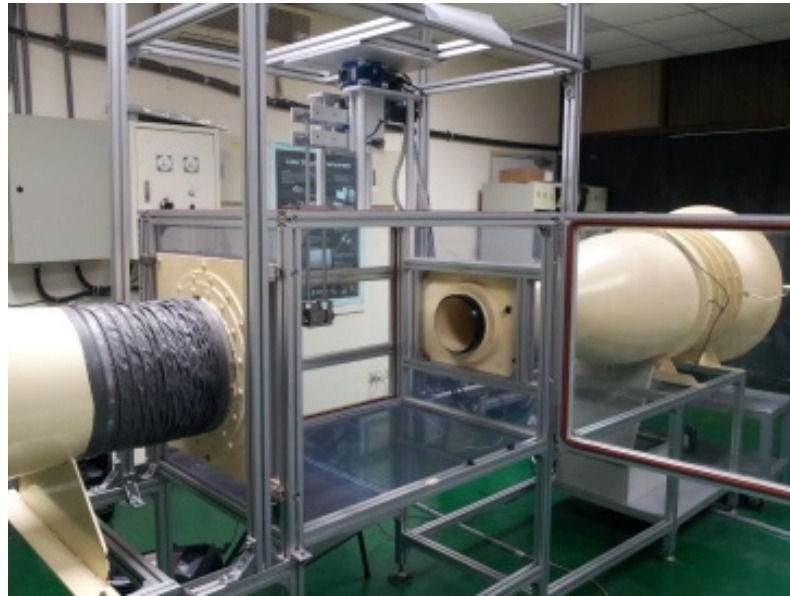
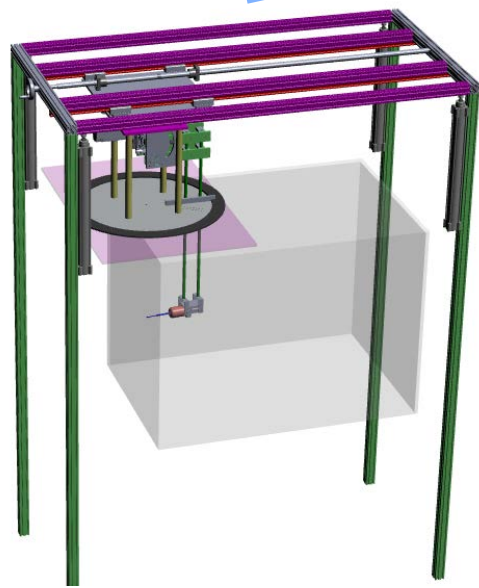
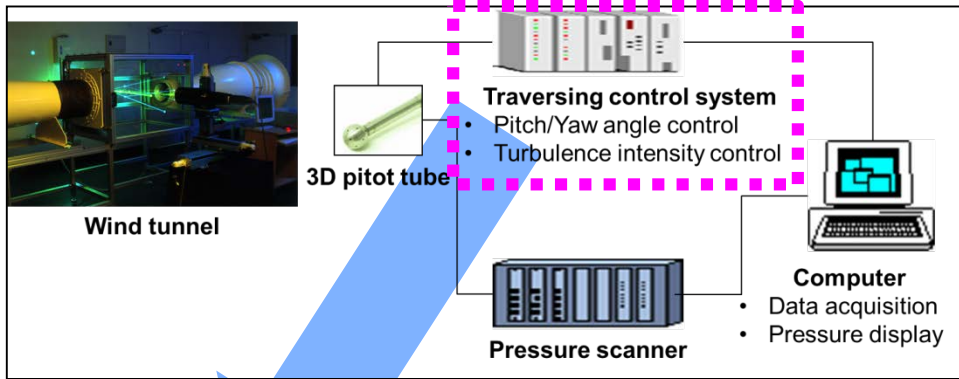


Grid-generated turbulent flow



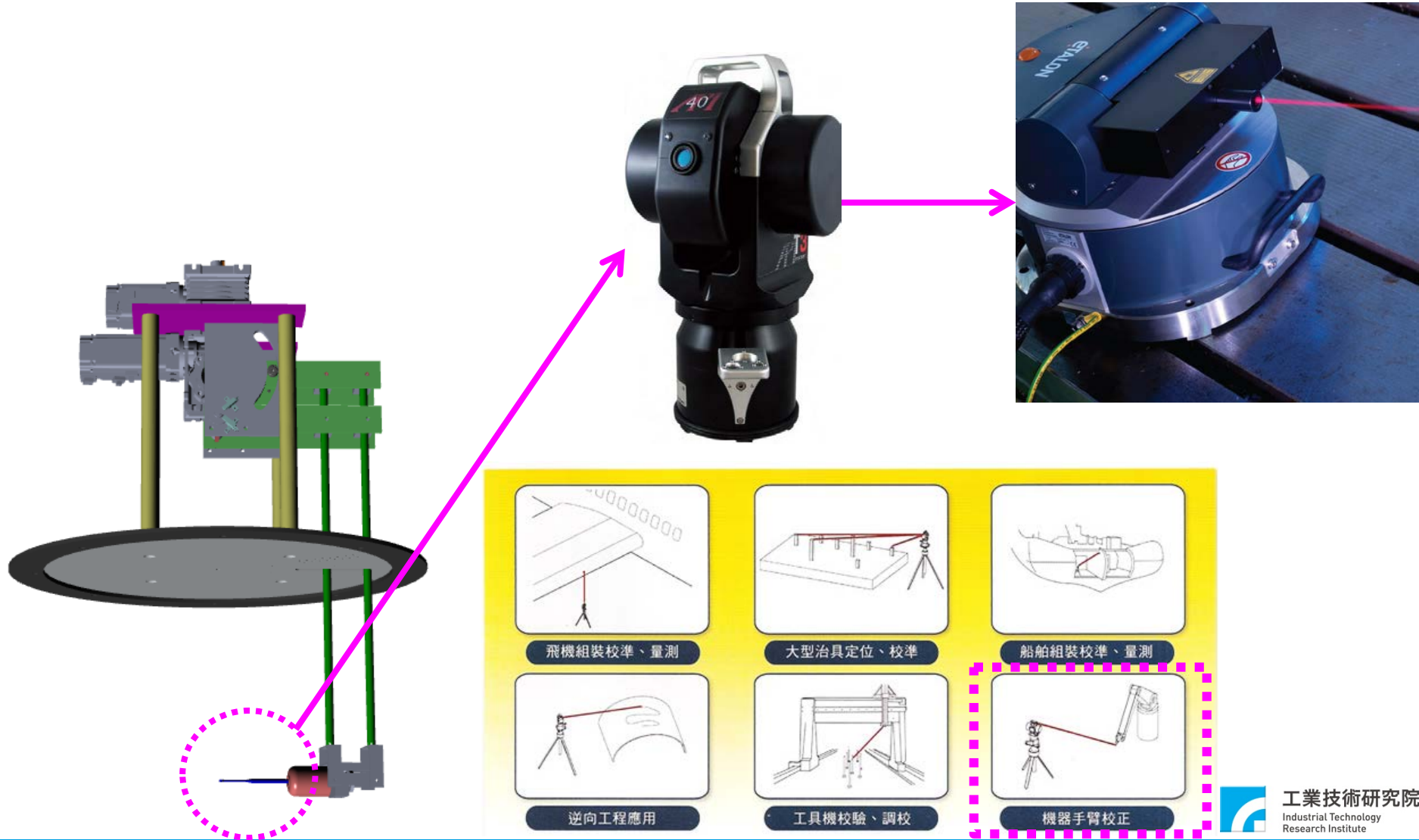
Calibration Facilities

- Traverse stage



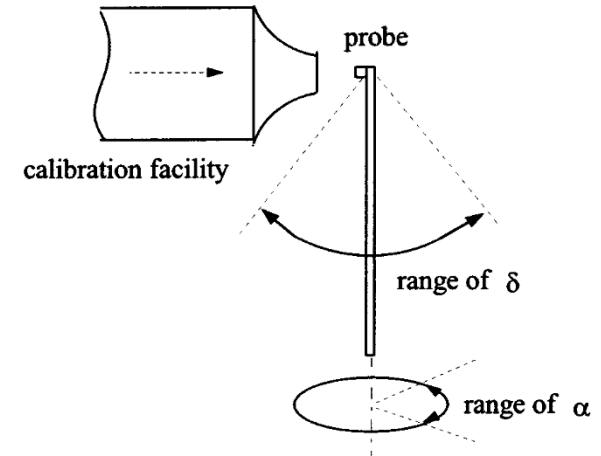
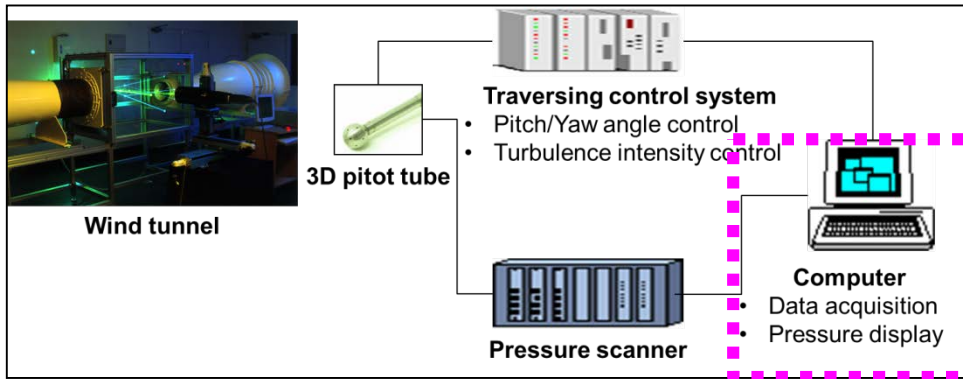
Calibration Facilities

- Angle measurements



Calibration Data Analysis

- Definition of pressure coefficients



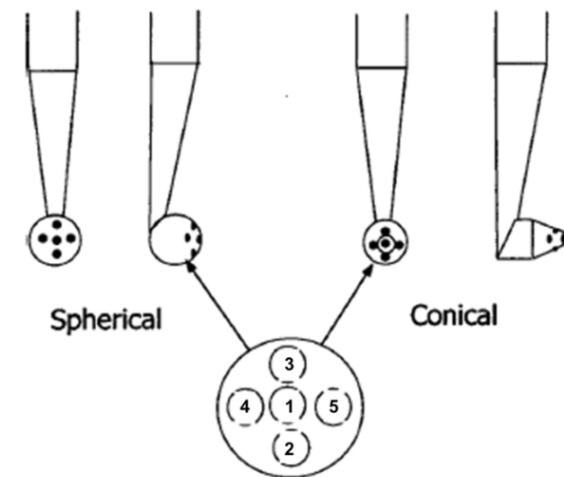
Input:

$$B_{\alpha} = \frac{p_4 - p_5}{Q'}, \quad B_{\delta} = \frac{p_1 - p_3}{Q'}$$

where $Q' = p_2 - 0.25 \times (p_1 + p_3 + p_4 + p_5)$.

Output:

$$\alpha, \delta, A_t = \frac{p_2 - p_t}{Q'}, \quad A_s = \frac{p_2 - p_5}{Q'}$$



Calibration Data Analysis

- Nulling method

Step 1: Align the probe so that the center hole is pointing towards a reference position.

Step 2: Rotate probe until $P_2=P_3$. This is the Yaw angle.

Step 3: Calculate Pitch Angle Pressure Coefficient $[(P_4-P_5)/(P_1-P_2)]$.

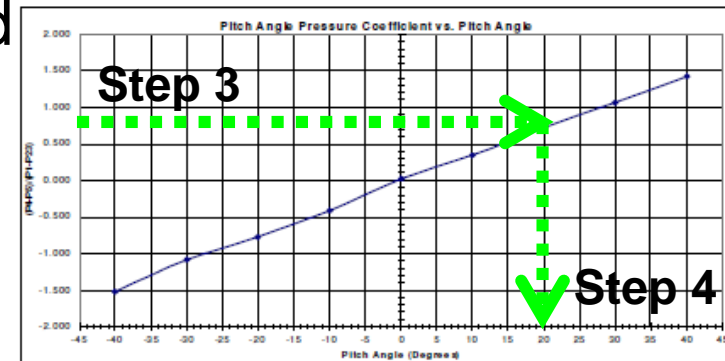
Step 4: Determine Pitch Angle.

Step 5: Determine Velocity Pressure Coefficient $[(P_t-P_s)/(P_1-P_2)]$.

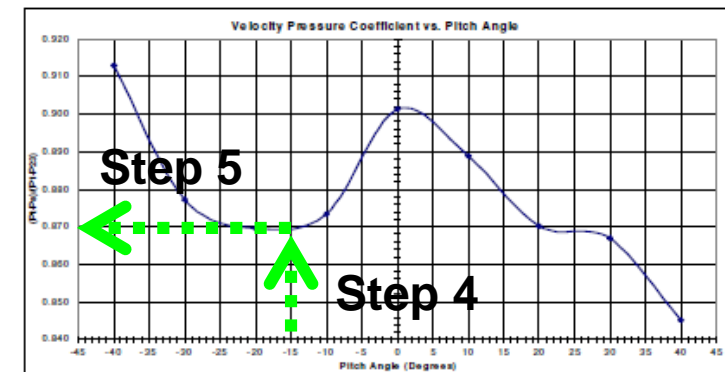
Step 6: Calculate Velocity pressure (P_t-P_s) .

Step 7: Determine Total Pressure Coefficient $[(P_1-P_t)/(P_t-P_s)]$.

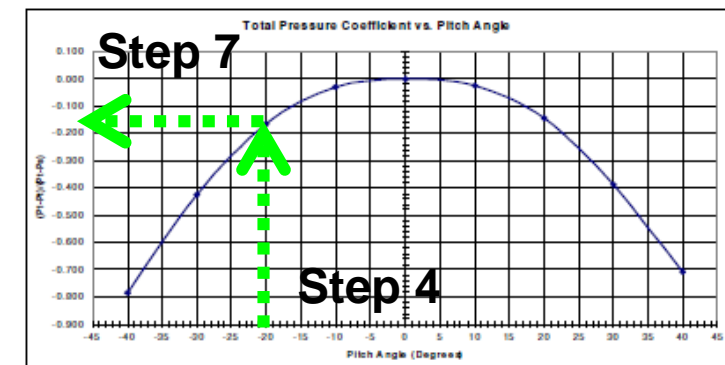
Step 8: Calculate (P_1-P_t) and obtain P_t .



Pitch angle vs. Pitch angle pressure coefficient



Pitch angle vs. Velocity pressure coefficient



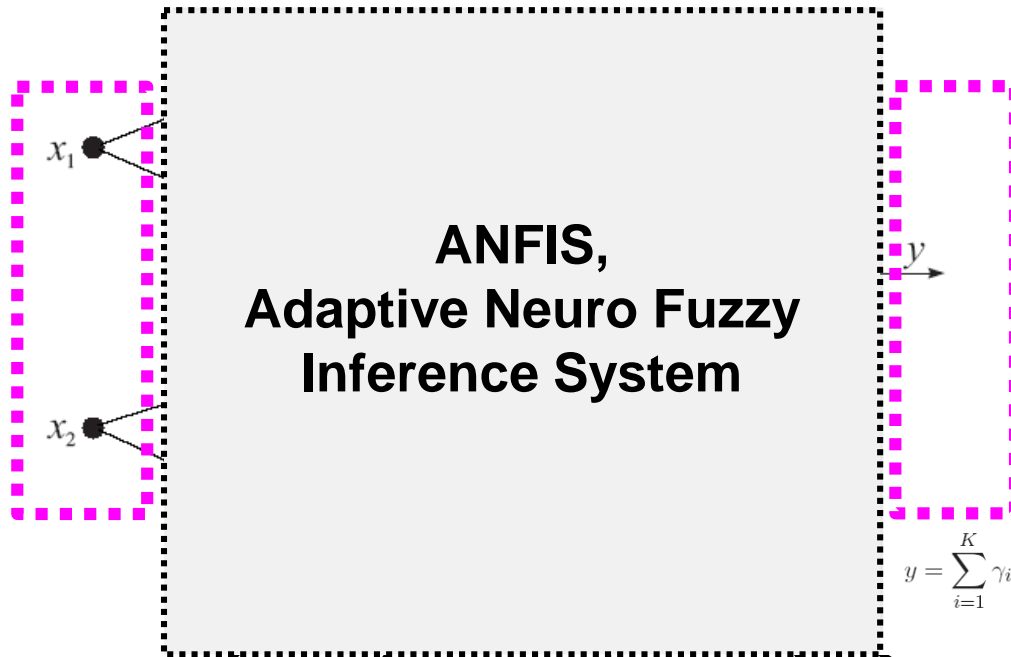
Pitch angle vs. Total pressure coefficient

Calibration Data Analysis

- Non-nulling method

INPUT

- Yaw and Pitch angle pressure coefficients



OUTPUT

- Yaw angle
- Pitch angle
- Total and Static pressure coefficient

$$y = \sum_{i=1}^K \gamma_i(\mathbf{x}) b_i \quad \text{with} \quad \gamma_i(\mathbf{x}) = \frac{\prod_{j=1}^p \exp\left(-\frac{(x_j - c_{ij})^2}{2\sigma_{ij}^2}\right)}{\sum_{i=1}^K \prod_{j=1}^p \exp\left(-\frac{(x_j - c_{ij})^2}{2\sigma_{ij}^2}\right)}$$

Nodes in this layer contains membership functions

This layer chooses the minimum value of two input weights

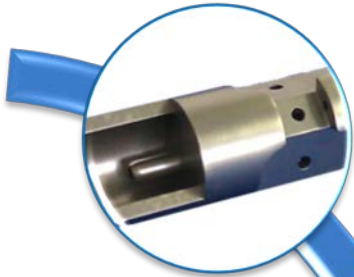
Every node of these layers calculates the weight, which is normalized

This layer includes linear functions, which are functions of the input signals

Compute the overall output by summing all incoming signals

Future Work

Calibration of 3D Pitot Tubes and Flow Measurements of Smokestack Emissions



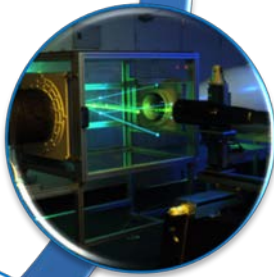
Pitot tube characterization

- Characterization of yaw and pitch angle for different types of 3D pitot tubes
- Flow visualization by CFD and PIV



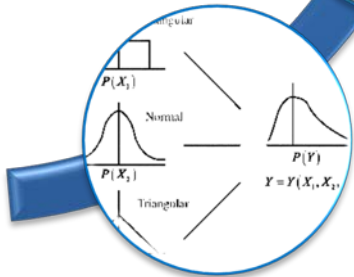
Standard traceability

- Integration of wind tunnel and traverse stage for 3D pitot tube calibration



Calibration method and facility establishment

- Comparison of calibration methods (nulling and non-nulling method)



Uncertainty evaluation

- Evaluation of uncertainty evaluation and calibration procedure



**Thanks for Your Listening &
Let's Work Together for a Better World**